# Betz Limit Derivation <br> Rev:20231228_1900 

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## GIVEN:



FIG. 1. Idealized stationary wind turbine (ground reference frame).

An idealized wind turbine fixed to the ground is extracting constant power, $P_{\text {net }}(W)$, from a steady and uniform wind having constant speed equal to $V_{\text {wind }}(\mathrm{m} / \mathrm{s})$. The density of the air is $\rho\left(\mathrm{kg} / \mathrm{m}^{3}\right)$, and the swept area of the rotor disk is $S\left(m^{2}\right)$. The turbine slows the airflow down by an amount equal to $\Delta V(\mathrm{~m} / \mathrm{s})$, such that the air speed far downstream is reduced to $V_{\text {wind }}-\Delta V$. This negative acceleration of the airflow results in a thrust force of the air acting on the rotor in the direction of the wind equal to $\vec{F}_{\mathrm{t}}(N)$. See figure 1. Additional parameters are given as:

$$
\begin{align*}
a & =\frac{1}{2} \frac{\Delta V}{V_{\text {wind }}} & & \text {; Dimensionless "Axial Induction Factor." }  \tag{1}\\
V & =V_{\text {wind }}-\frac{1}{2} \Delta V & & ; \text { Speed of the air passing through the rotor disk }(\mathrm{m} / \mathrm{s}) .^{\text {a }}  \tag{2}\\
\dot{m} & =\rho S V & & \text {; Mass flow rate of air passing through the rotor disk }(\mathrm{kg} / \mathrm{s}) . \tag{3}
\end{align*}
$$

[^0]
## FIND:

Part A: Derive dimensionless equations, each written as a function of the axial induction factor $a$, for: 1.) the coefficient of thrust force of the air acting on the rotor disk, $C F_{\mathrm{t}}=f(a)$; and 2.) the coefficient of net power extracted from the wind, $C P_{\text {net }}=f(a)$.

Part B: Determine: 1.) the maximum possible net power coefficient, $C P_{\text {netmax }}$; and 2.) the axial induction factor value, $a_{\text {netmax }}$, required to maximize the net power. The reference constants to be used for normalizing the derived equations into dimensionless form are given as:

$$
\begin{array}{ll}
P_{\text {wind }}=\frac{1}{2} \rho S V_{\text {wind }}^{3} & ; \text { Total kinetic power of the wind passing through area } \mathrm{S}(W) .{ }^{\mathrm{b}} \\
F_{\text {wind }}=\frac{1}{2} \rho S V_{\text {wind }}^{2} & ; \text { Force which satisfies the equation: } P_{\text {wind }}=F_{\text {wind }} V_{\text {wind }}(N) .{ }^{\mathrm{b}} \tag{5}
\end{array}
$$

Assumptions:

- The air behaves as an ideal Newtonian fluid; it is incompressible and inviscid.
- The airflow is steady, laminar and adiabatic.
- The rotor acts as an ideal $100 \%$ efficient actuator disk with zero losses.
- Losses due to swirl in the outflow and work done on the air outside the stream tube are negligible.
- The flow is considered to be axial with uniform velocity within any cross sectional area slice of the stream tube normal to the x axis.
- A $100 \%$ efficient transmission connects the rotor to a $100 \%$ efficient electrical generator that provides $P_{\text {net }}$ power output to a connected load.

[^1]
## SOLUTION:



FIG. 2. Free body diagram of air passing through control volume (ground/rotor reference frame.)

In time $\Delta t$, a discreet mass of air located far upstream from the rotor disk, $\Delta m$, having constant speed, $V_{1}=V_{\text {wind }}$, enters the control volume where negative acceleration is taking place. During the same $\Delta t$ time period, an equal mass of air located far downstream from the rotor disk, $\Delta m$, exits the control volume with constant lesser speed, $V_{2}=V_{\text {wind }}-\Delta V$. See figure 2. The effective reduction in speed of mass $\Delta m$ from $V_{1}$ upstream to $V_{2}$ downstream, $\Delta V$, is caused by the thrust reaction force of the rotor acting on the air inside the control volume, $-\vec{F}_{\mathrm{t}}$. Applying Newton's second law, $\vec{F}=m \vec{a},{ }^{\text {c }}$ and substituting the given equation for mass flow rate, $\dot{m}=\rho S V$, we get an equation for the thrust force $F_{\mathrm{t}}$ :

$$
\begin{align*}
-\vec{F}_{\mathrm{t}} & =\Delta m \vec{a} & & ; \text { Newton's 2nd. Substitute average acceleration, } \vec{a}=\overrightarrow{\Delta V} / \Delta t:^{\mathrm{c}}  \tag{6}\\
-\vec{F}_{\mathrm{t}} & =\Delta m \frac{\Delta \vec{V}}{\Delta t} & & ; \text { Both }-\vec{F}_{\mathrm{t}} \text { and } \overrightarrow{\Delta V} \text { vectors have negative direction, so: }  \tag{7}\\
-F_{\mathrm{t}} & =\Delta m\left(\frac{-\Delta V}{\Delta t}\right) & & ; \text { Converted from vector to scaler. Multiply both sides by }-1 \text { : }  \tag{8}\\
F_{\mathrm{t}} & =\Delta m \frac{\Delta V}{\Delta t} & & ; \text { Rearrange: }  \tag{9}\\
& =\frac{\Delta m}{\Delta t} \Delta V & & ; \text { Thrust force is mass flow rate } \dot{m} \text { times } \Delta V .  \tag{10}\\
& =\dot{m} \Delta V & & ; \text { Substitute given equation, } \dot{m}=\rho S V:  \tag{11}\\
& =\rho S V \Delta V & & ; \text { Rearrange: }  \tag{12}\\
F_{\mathrm{t}} & =\rho S \Delta V V & & \text { Thrust force of air acting on rotor, } F_{\mathrm{t}}=f(V, \Delta V) . \tag{13}
\end{align*}
$$

[^2]Since all components of the turbine are $100 \%$ efficient, the net power output from the turbine, $P_{\text {net }}$, is equal to the power transferred from the air into the rotor, $P_{\mathrm{ar}}$. Applying the general vector equation for power, $P=\vec{F} \cdot \vec{V}$, to the force acting on the air passing through the rotor disk, we get an equation for the net power:

$$
\begin{align*}
P_{\mathrm{ra}} & =-\vec{F}_{\mathrm{t}} \cdot \vec{V} & & ; \text { Power transferred from rotor into the air; vector form. }  \tag{14}\\
P_{\mathrm{ra}} & =-F_{\mathrm{t}} V & & ; \text { Scalar form. Negative power. But } P_{\mathrm{ra}}=-P_{\mathrm{ar}}, \text { so: }  \tag{15}\\
P_{\mathrm{ar}} & =F_{\mathrm{t}} V=P_{\mathrm{net}} & & ; \text { Power transferred from air into the rotor. Substitute } F_{\mathrm{t}}:  \tag{16}\\
P_{\mathrm{net}} & =\rho S \Delta V V^{2} & & ; \text { Net wind power extracted by turbine, } P_{\text {net }}=f(V, \Delta V) . \tag{17}
\end{align*}
$$

Substituting the given equation for $V$ into the equations for $F_{\mathrm{t}}$ and $P_{\text {net }}$ yields:

$$
\begin{align*}
V & =V_{\text {wind }}-\frac{1}{2} \Delta V & & ; \text { Given equation for } V . \text { Substitute into } F_{\mathrm{t}} \text { and } P_{\mathrm{net}}:  \tag{18}\\
F_{\mathrm{t}} & =\rho S \Delta V\left(V_{\text {wind }}-\frac{1}{2} \Delta V\right) & & ; \text { Thrust force of air on rotor, } F_{\mathrm{t}}=f\left(V_{\text {wind }}, \Delta V\right), \text { and }  \tag{19}\\
P_{\mathrm{net}} & =\rho S \Delta V\left(V_{\text {wind }}-\frac{1}{2} \Delta V\right)^{2} & & ; \text { Net power extracted by turbine, } P_{\mathrm{net}}=f\left(V_{\text {wind }}, \Delta V\right) . \tag{20}
\end{align*}
$$

To normalize the equations into dimensionless form, we solve the given axial induction factor equation for $\Delta V$, and substitute that term into the equations. First for $F_{\mathrm{t}}$ :

$$
\begin{align*}
a & =\frac{1}{2} \frac{\Delta V}{V_{\text {wind }}} & & ; \text { Given equation for } a . \text { Re-arrange: }  \tag{21}\\
\Delta V & =2 a V_{\text {wind }} & & ; \Delta V=f\left(V_{\text {wind }}, a\right) . \text { Substitute into } F_{\mathrm{t}} \text { equation: }  \tag{22}\\
F_{\mathrm{t}} & =\rho S\left(2 a V_{\text {wind }}\right)\left(V_{\text {wind }}-\frac{1}{2}\left(2 a V_{\text {wind }}\right)\right) & & ; F_{\mathrm{t}}=f\left(V_{\text {wind }}, a\right) . \text { Gather } V_{\text {wind }} \text { terms: }  \tag{23}\\
& =\rho S V_{\text {wind }}^{2} 2 a(1-a) & & ; \text { Cleverly multiply and divide by } 2:  \tag{24}\\
& =\frac{1}{2} \rho S V_{\text {wind }}^{2} \cdot 4 a(1-a) & & ; \text { Notice that first term is equal to } F_{\text {wind }}, \text { so: }  \tag{25}\\
F_{\mathrm{t}} & =F_{\text {wind }} \cdot C F_{\mathrm{t}} & & ; \text { Generic equation for rotor thrust force, } F_{\mathrm{t}}, \text { where: }  \tag{26}\\
{\left[C F_{\mathrm{t}}\right.} & =4 a(1-a)] & & ; \text { Answer A1: } C F_{\mathrm{t}}=f(a) . \tag{27}
\end{align*}
$$

Next substitute $\Delta V=2 a V_{\text {wind }}$ into $P_{\text {net }}=f\left(V_{\text {wind }}, \Delta V\right)$ :

$$
\begin{align*}
P_{\mathrm{net}} & =\rho S\left(2 a V_{\mathrm{wind}}\right)\left(V_{\mathrm{wind}}-\frac{1}{2}\left(2 a V_{\text {wind }}\right)\right)^{2} & & ; P_{\mathrm{net}}=f\left(V_{\text {wind }}, a\right) . \text { Gather } V_{\text {wind }} \text { terms: }  \tag{28}\\
& =\rho S V_{\mathrm{wind}}^{3} 2 a(1-a)^{2} & & ; \text { Multiply and divide by } 2:  \tag{29}\\
& =\frac{1}{2} \rho S V_{\mathrm{wind}}^{3} \cdot 4 a(1-a)^{2} & & ; \text { Notice that first term is equal to } P_{\text {wind }}, \text { so: }  \tag{30}\\
P_{\mathrm{net}} & =P_{\text {wind }} \cdot C P_{\mathrm{net}} & & ; \text { Generic equation for net power, } P_{\mathrm{net}}, \text { where: }  \tag{31}\\
{\left[C P_{\mathrm{net}}\right.} & \left.=4 a(1-a)^{2}\right] & & ; \text { Answer A2: } C P_{\mathrm{net}}=f(a) . \tag{32}
\end{align*}
$$

Figure 3 shows a plot of CFt and CPnet. To determine the maximum possible net power coefficient, we take the derivative of the $C P_{\text {net }}=f(a)$ equation with respect to $a$, set the resulting equation equal to zero, then solve for the


FIG. 3. Plot of functions $C F_{\mathrm{t}}$ and $C P_{\text {net }}$ vs Axial Induction Factor $a$.
roots:

$$
\begin{align*}
& C P_{\text {net }}=4 a(1-a)^{2}  \tag{33}\\
& =4 a(1-a)(1-a)  \tag{34}\\
& =4 a\left(1-2 a+a^{2}\right)  \tag{35}\\
& C P_{\text {net }}=4 a^{3}-8 a^{2}+4 a  \tag{36}\\
& \text {; } C P_{\text {net }} \text { in polynomial form. Take the derivative: } \\
& \text {; Derivative of } C P_{\text {net }} \text {. Set equal to } 0 \text { and divide by } 4 \text { : }  \tag{37}\\
& \text {; Roots of this equation are } a_{\text {netmax }} \text { and } a_{\text {netmin }} \text { of } C P_{\text {net }} \text {. }  \tag{38}\\
& \text { roots }=\frac{-B \pm \sqrt{B^{2}-4 A C}}{2 A}  \tag{39}\\
& \text {; Quadratic equation. Use: } A=3, B=-4 \text { and } C=1 \text { : } \\
& =\frac{-(-4) \pm \sqrt{(-4)^{2}-4(3)(1)}}{2(3)}  \tag{40}\\
& =\frac{4 \pm \sqrt{16-12}}{6} \text {; }  \tag{41}\\
& =\frac{1}{6}(4 \pm 2) \quad ;  \tag{42}\\
& =\frac{1}{3}(2 \pm 1) \quad ;  \tag{43}\\
& \text { roots }=1, \frac{1}{3} \quad \text {; Root values: } 1=a_{\text {netmin }} \text { and } 1 / 3=a_{\text {netmax }} .  \tag{44}\\
& {\left[a_{\text {netmax }}=\frac{1}{3}\right] \quad \text {; Answer B2: Value of } a \text { which maximizes } C P_{\text {net }} \text {. }} \tag{45}
\end{align*}
$$

Plugging $a_{\text {netmax }}$ into $C P_{\text {net }}=f(a)$ reveals the maximum possible net power coefficient:

$$
\begin{align*}
C P_{\text {net }} & =4 a(1-a)^{2} & & ; C P_{\text {net }}=f(a) . \text { Plug in } a_{\text {netmax }}=1 / 3:  \tag{46}\\
C P_{\text {netmax }} & =4\left(\frac{1}{3}\right)\left(1-\frac{1}{3}\right)^{2} & & ;  \tag{47}\\
& =\frac{4}{3}\left(\frac{2}{3}\right)^{2} & & ;  \tag{48}\\
& =\frac{4}{3}\left(\frac{4}{9}\right) & & ; \text { Answer B1: } C P_{\text {netmax }}=16 / 27=\text { Betz Limit. } \tag{49}
\end{align*}
$$


[^0]:    a It is generally accepted that the velocity of the air as it passes through an (idealized) actuator disk, $V$, (for a propeller or turbine rotor,) has a magnitude halfway between the far upstream velocity, $V_{1}$, and the far downstream velocity, $V_{2}$, i.e. $V=1 / 2\left(V_{1}+V_{2}\right)$. The mathematical proof of this concept is not provided in this paper.

[^1]:    ${ }^{\text {b }} V_{\text {wind }}, P_{\text {wind }}$ and $F_{\text {wind }}$, are the reference constants used for converting the dimensional equations for velocity, power and force into dimensionless form. The $P_{\text {wind }}$, "Power of the wind" constant, represents the kinetic power that a wind flowing freely through a cross sectional area $S$ possesses, as measured in the ground reference frame. The $F_{\text {wind }}$, "Force of the wind" constant, represents the (theoretical) force that would be required to "stop the wind" and extract the full power from the wind.

[^2]:    c In this article, the symbol: "a" is used to represent two different parameters: 1.) the axial induction factor, and 2.) the acceleration term in Newton's second law equation, $(\vec{F}=m \vec{a})$. Hopefully, this will not cause any confusion since the acceleration term is used only briefly here.

